This Page Is Inserted by IFW Operations and is not a part of the Official Record

BEST AVAILABLE IMAGES

Defective images within this document are accurate representations of the original documents submitted by the applicant.

Defects in the images may include (but are not limited to):

- BLACK BORDERS
- TEXT CUT OFF AT TOP, BOTTOM OR SIDES
- FADED TEXT
- ILLEGIBLE TEXT
- SKEWED/SLANTED IMAGES
- COLORED PHOTOS
- BLACK OR VERY BLACK AND WHITE DARK PHOTOS
- GRAY SCALE DOCUMENTS

IMAGES ARE BEST AVAILABLE COPY.

As rescanning documents will not correct images, please do not report the images to the Image Problem Mailbox.

THIS PAGE BLANK (USFIO)



INTERNATIONAL APPLICATION PUBLISHED UNDER THE PATENT COOPERATION TREATY (PCT)

(51) International Patent Classification 6:

F41H 5/04

A1

(11) International Publication Number:

WO 98/37376

(43) International Publication Date:

27 August 1998 (27.08.98)

(21) International Application Number:

PCT/US98/03128

(22) International Filing Date:

18 February 1998 (18.02.98)

(30) Priority Data:

08/803,040

19 February 1997 (19.02.97) US

(71) Applicant: UNITED DEFENSE, L.P. [US/US]; Office of General Counsel, Suite 700, 1525 Wilson Boulevard, Arlington, VA 22209 (US).

(71)(72) Applicant and Inventor: MOORE, Franklin, C. [US/US]; 215 South Thistle Lane, Maitland, FL 32751 (US).

(72) Inventors: SCHADE, David, Arthur; 8 Cranfield Avenue, San Carlos, CA 94070 (US). MCARTHUR, Michael, Brian; 14629 Branham Lane, San Jose, CA 95124 (US). PIKE, Torrey, Lane; 155 Lawnview Circle, Danville, CA 94526 (US). THOMAS, George, Edgar, Jr.; 161 S. California Avenue #K-204, Palo Alto, CA 94306 (US). MIDDIONE, Mark, Albert; 16 Cooper's Hawk Court, Scotts Valley, CA 95066 (US).

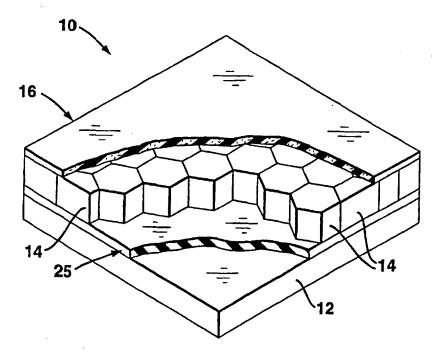
(74) Agent: CALDWELL, John, W.; Woodcock Washburn Kurtz Mackiewicz & Norris LLP, 46th floor, One Liberty Place, Philadelphia, PA 19103 (US).

(81) Designated States: AL, AM, AT, AU, AZ, BA, BB, BG, BR, BY, CA, CH, CN, CU, CZ, DE, DK, EE, ES, FI, GB, GE, GH, GM, GW, HU, ID, IL, IS, JP, KE, KG, KP, KR, KZ, LC, LK, LR, LS, LT, LU, LV, MD, MG, MK, MN, MW, MX, NO, NZ, PL, PT, RO, RU, SD, SE, SG, SI, SK, SL, TJ, TM, TR, TT, UA, UG, UZ, VN, YU, ZW, ARIPO patent (GH, GM, KE, LS, MW, SD, SZ, UG, ZW), Eurasian patent (AM, AZ, BY, KG, KZ, MD, RU, TJ, TM), European patent (AT, BE, CH, DE, DK, ES, FI, FR, GB, GR, IE, IT, LU, MC, NL, PT, SE), OAPI patent (BF, BJ, CF, CG, CI, CM, GA, GN, ML, MR, NE, SN, TD, TG).

Published

With international search report. With amended claims and statement.

(54) Title: COMPOSITE ARMOR FOR A VEHICLE AND METHOD FOR MANUFACTURING SUCH AN ARMOUR



(57) Abstract

The invention provides an improved composite armor system that may be used for an upper hull for a composite armor vehicle. The invention provides a complete composite armor system. A composite armor support layer is formed by the hull of the vehicle or protective system. Armor tiles are attached to the composite armor support layer to improve the armor protection. A signature layer is then applied.

FOR THE PURPOSES OF INFORMATION ONLY

Codes used to identify States party to the PCT on the front pages of pamphlets publishing international applications under the PCT.

		ES	Spain	LS	Lesotho	SI	Slovenia
AL.	Albania	FI	Finland	L'T	Lithuania	SK	Slovakia
AM	Armenia	FR	France	LU	Luxembourg	SN	Senegal
AT	Austria	GA	Gabon	LV	Latvia	SZ	Swaziland
ΑU	Australia	GB	United Kingdom	MC	Monaco	TD	Chad
AZ	Azerbaijan	GE	Georgia	MD	Republic of Moldova	TG	Togo
BA	Bosnia and Herzegovina	GII	Ghana	MG	Madagascar	TJ	Tajikistan
BB	Barbados	GN	Guinea	MK	The former Yugoslav	TM	Turkmenistan
BE	Belgium	GR-	Greece		Republic of Macedonia	TR	Turkey
BI	Burkina Faso	HU	Hungary	ML	Mali	TT	Trinidad and Tobago
BG	Bulgaria	IE	Ireland	MN	Mongolia	UA	Ukraine
BJ	Benin		Israel	MR	Mauritania	UG	Uganda
BR	Brazil	II.	Iceland	MW	Malawi	US	United States of America
BY	Belarus	IS		MX	Mexico	UZ	Uzbekistan
CA	Canada	IT	Italy	NE.	Niger	VN	Viet Nam
CF	Central African Republic	JP	Japan	NI.	Netherlands	YU	Yugoslavia
CG	Congo	KE	Kenya	NO	Norway	ZW	Zimbabwe
CH	Switzerland	KG	Kyrgyzstan	NZ	New Zealand		
CI	Côte d'Ivoire	KP	Democratic People's		Poland		
CM	Cameroon		Republic of Korea	PL			
CN	China	KR	Republic of Korea	PT	Portugal		
cu	Cuba	KZ	Kazakstan	RO	Romania		
C%	Czech Republic	LC	Saint Lucia	RU	Russian Federation		
DE	Germany	LI	Liechtenstein	SD	Sudan		
DK	Denmark	LK	Sri Lanka	SE	Sweden		
EE	Estonia	LR	Liberia	SG	Singapore		

WO 98/37376 PCT/US98/03128

COMPOSITE ARMOR FOR A VEHICLE AND METHOD FOR MANUFACTURING SUCH AN ARMOUR

5 BACKGROUND OF THE INVENTION

The present invention relates generally to composite armor used on an upper hull of a vehicle.

In the prior art, composite armor vehicles did not provide sufficient protection.

10

15

20

25

SUMMARY OF THE INVENTION

It is an object of the invention to provide a composite armor which provides sufficient protection to the upper hull of a composite armor vehicle.

It is another object of the invention to provide ceramic tile armor on a composite vehicle that can sustain multiple hits.

The invention provides an improved method and apparatus for creating a composite armor package. The invention uses a toughened epoxy adhesive and a rubber interlayer to improve the bonding of ceramic tile to armor and provides a means of providing a desired radar signature.

BRIEF DESCRIPTION OF THE DRAWINGS

Figure 1 is a cut away perspective view of a preferred embodiment of the invention.

Figure 2 illustrates a detailed perspective view of the attachment layer illustrated in Figure 1.

Figure 3 is a schematic view of the signature layer illustrated 30 in Figure 1.

DESCRIPTION OF THE PREFERRED EMBODIMENTS

A armor system 10, as shown in Figure 1, may be used on a composite armor military vehicle or for other types of composite armor

15

20

25

35

,

protection. A armor system 10, comprises a composite support layer 12, an interlayer 25, a layer of ceramic tiles 14, and a signature layer 16.

In the preferred embodiment, the composite support layer 12 is made of fiberglass, preferably S-2 glass. The S-2 glass is unidirectional fiberglass tape (unitape) layer between 0.75 inches and 1 inch thick. In other embodiments, the fiberglass may use a bidirectional woven fiberglass piece.

Figure 2 illustrates a detailed perspective view of the interlayer 25 illustrated in Figure 1. The interlayer 25, comprises a first layer of epoxy 27 between a first side of the composite support layer 12 and a first side of a layer of EPDM rubber 28, a first scrim cloth 29, which comprises an open weave of fiberglass material embedded in the first layer of epoxy 27, a second layer of epoxy 30 between a second side of the layer of EPDM rubber 28 and a first side of the layer of ceramic tiles 14, and a second scrim cloth 31, which comprises an open weave of fiberglass material embedded in the second layer of epoxy 30. The ceramic tiles 14 are of an armor ceramic. In the preferred embodiment the armor ceramic of the layer of ceramic tiles 14 are Cercom hot pressed silicon carbide Type N tiles made by Cercom Incorporated at 1960 Watson Way, Vista, California 92063. The signature layer 16 is place over the layer of ceramic tiles 14 and the edges of the armor. The signature layer 16 is described in more detail below.

In the manufacture of the preferred embodiment the composite support layer 12 is first prepared by removing all oil and grease from the surface of the composite support layer, and then by sand blasting or grinding the surface to be bonded. The first and second sides of the layer of EPDM rubber 28 are sanded with 80-grit paper, so that the surface gloss is removed. The ceramic tiles of the layer of ceramic tiles 14 are cleaned. The layer of ceramic tiles 14, the layer of EPDM rubber 28, and the composite support layer 12 are primed. The primer promotes surface bonding with the adhesive. SC-11 epoxy, which is a toughened epoxy made by Applied Polymeric Incorporated (API) at 850 Teal Drive, Benicia, California 94510, is mixed and 5% by weight of fumed silica is mixed into the epoxy to thicken the epoxy and make it easier to work with. The epoxy is spread over the first side of the composite support layer 12. The first scrim cloth 29 is placed on the epoxy spread over the first side of the

10

15

composite support layer 12. Epoxy is spread over the first scrim cloth 29, thus forming the first layer of epoxy 27 with the first scrim cloth 29 embedded therein. The first scrim cloth 29 serves as a wick and spacer to uniformly distribute the epoxy between the layer of EPDM rubber 28 and the ceramic tiles 14. The layer of EPDM rubber 28 is placed on the first layer of epoxy 27. Epoxy is spread over the second side of the layer of EPDM rubber 28. The second scrim cloth 31 is placed over the epoxy spread over the second side of the layer of EPDM rubber 28. Epoxy is spread over the second scrim cloth 31, thus forming the second layer of epoxy 30 with the second scrim cloth 31 embedded therein. The layer of ceramic tiles 14 is placed on the second layer of epoxy 30. Epoxy is placed over the tiles to fill in gaps, and allowed to cure over night. The dimension of the gap between the tiles is between 0.010 inches and 0.10 inches. When protection between multiple hits is desired a gap of 0.015 inches is preferred. When protection from small threats is desired a gap of 0.080 is preferred.

Figure 3 is a schematic view of the signature layer 16, which is a radar attenuating layer. A layer of epoxy is placed on the layer of ceramic tiles 14. A ground plane scrim 21, formed by a tightly woven graphite cloth, is placed on the layer of epoxy on the layer of ceramic tiles 20 14. A layer of epoxy is placed on the ground plane scrim 21. Plastic is placed over the epoxy on the ground plane scrim 21 and a vacuum is used to draw the epoxy into the ground plane scrim 21. The plastic is then removed from the epoxy. A first plurality of layers of fiberglass 22 impregnated with epoxy and a radar attenuating material manufactured 25 by Lockheed/Martin are placed over the tightly ground plane scrim 21. The type of radar attenuating material and type and number of layers of fiberglass are dependent on the desired radar signature. Other types of radar attenuating layers may be used for the signature layer. Such radar attenuating layers are known in the prior art. In the preferred 30 embodiment the first plurality of layers of fiberglass 22 are six layers of 977 fiberglass, which has a tight weave and thin fibers. A second plurality of layers of fiberglass 23 impregnated with epoxy, but not treated with radar attenuating material are placed over the first plurality of layers of fiberglass 22. In the preferred embodiment the second plurality of layers 35 of fiberglass 23 are three layers of 977 fiberglass, which has a tight weave

10

15

20

25

30

and thin fibers. The signature layer 16 comprises the ground plane scrim 21, the first plurality of layers of fiberglass 22 and the second plurality of layers of fiberglass 23.

The composite support layer 12 forms part of a composite military vehicle or a building, where protection from munitions is desirable. When the armor system 10 in the preferred embodiment was hit with a projectile, the inventive armor minimized ceramic tile displacement, by limiting the tiles removed or damaged by the projectile to a small area.

In other embodiments, other types of adhesives may used instead of SC-11 epoxy. Other elastomer materials may be used in place of EPDM rubber replacing the layer of EPDM rubber with an elastomer layer. Other embodiments may leave out the scrim cloth or use a scrim cloth of a different material. Other types of interlayers may be used to bind the ceramic tiles 14 to the composite support layer 12, such as replacing the EPDM rubber with fiberglass. The ceramic tiles 14 may be made of other armor ceramics. Armor ceramics are made of a high density, high strength crystalline material, such as alumina. In some armor ceramics, the crystalline material is bonded with an amorphous substance such as glass.

The wrapping of the signature layer around the edges of the composite support structure, signifies that edges of the composite support structure are wrapped to provide the desired radar signature.

The use of a composite support layer instead of a metallic support layer, provides inherent spall protection and a light weight inexpensive armor. Therefore the invention is a light weight inexpensive armor that is able to withstand many conventional threats and has radar attenuation properties and inherent spall protection in a single package.

While preferred embodiment of the present invention has been shown and described herein, it will be appreciated that various changes and modifications may be made therein without departing from the spirit of the invention as defined by the scope of the appended claims.

What is claimed is:

1. An composite armor system, comprising: a composite support layer;

a plurality of tiles forming a layer of tiles with a first side and a second side;

an interlayer binding the layer of tiles to the composite support layer; and

a attenuating layer with a first side and a second side, wherein the first side of the signature layer is bonded to the second side of the layer of tiles.

2. The composite armor system, as claimed in claim 1, wherein the interlayer, comprises:

an elastomer layer with a first side and a second side with the first side of the elastomer layer adjacent to the first side of the composite support layer;

a first adhesive layer binding the first side of the composite support layer to the first side of the elastomer layer;

a second adhesive layer binding the second side of the elastomer layer with the first side of the layer of tiles; and

3. The composite armor system, as claimed in claim 2, wherein the signature layer, comprises:

a ground plane scrim with a first side and a second side, wherein the first side of the ground plane is bonded to the second side of the layer of tiles; and

a first plurality of layers of fiberglass impregnated with an epoxy with a first side and a second side, wherein the first side of the first plurality of layers of fiberglass are bonded to the second side of the ground plane scrim.

4. The composite armor system, as claimed in claim 3, wherein the signature layer, further comprises, a second plurality of layers of fiberglass with a first side and a second side, wherein the first side of the second plurality of layers of fiberglass is bonded to the second side of the first plurality of layers of fiberglass.

- 5. The composite armor system, as claimed in claim 4, wherein the first plurality of layers of fiberglass is impregnated with a radar attenuating material, and wherein the second plurality of layers of fiberglass is not impregnated with the radar attenuating material.
- 6. The composite armor system, as claimed in claim 5, wherein the tiles of the layer of tiles are made of an armor ceramic.
- 7. The composite armor system, as claimed in claim 6, wherein the ground plane scrim is a woven graphite cloth.
- 8. The armor system, as claimed in claim 7, further comprising:
 - a first scrim cloth embedded in the first adhesive layer; and a second scrim cloth embedded in the second adhesive layer.
- 9. The armor system, as claimed in claim 8, wherein the scrim cloth is a net of fiberglass material.
- 10. The armor system, as claimed in claim 9, wherein the elastomer layer is EPDM rubber.
- 11. The armor system, as claimed in claim 10, wherein the first adhesive layer and the second adhesive layer are epoxy layers.
- 12. The composite armor system, as claimed in claim 3, wherein the tiles of the layer of tiles are made of an armor ceramic.
- 13. The composite armor system, as claimed in claim 12, wherein the ground plane scrim is a woven graphite cloth.
- 14. The armor system, as claimed in claim 13, further comprising:
 - a first scrim cloth embedded in the first adhesive layer; and a second scrim cloth embedded in the second adhesive layer.

- 15. The armor system, as claimed in claim 14, wherein the scrim cloth is a net of fiberglass material.
- 16. The armor system, as claimed in claim 15, wherein the elastomer layer is EPDM rubber.
- 17. The composite armor system, as claimed in claim 1, wherein the tiles of the layer of tiles are made of an armor ceramic.
- 18. A method of manufacturing an armor system, comprising the steps of:

bonding a first side of an elastomer layer to a first side of a composite support layer;

bonding a first side of a layer of ceramic tiles to a second side of the elastomer layer; and

bonding a signature layer to a second side of the layer of ceramic tiles.

19. The method, as claimed in claim 18, wherein the step of bonding the signature layer, comprises the steps of:

bonding a first side of a ground plane scrim to the second side of the layer of tiles; and

bonding a first side of a first plurality of layers of fiberglass to a second side of the ground plane scrim.

20. The method, as claimed in claim 19, wherein the step of bonding the signature layer, further comprises the step of, bonding a first side of a second plurality of layers of fiberglass to a second side of the first plurality of layers of fiberglass, wherein the first plurality of layers of fiberglass is impregnated with a radar attenuating material and the second plurality of layers of fiberglass is not impregnated with the radar attenuating material.

AMENDED CLAIMS

[received by International Bureau on 15 July 1998 (15.07.98); original claims 1 and 2 amended; remaining claims unchanged (1page)]

- 1. An composite armor system, comprising:
 - a composite support layer;

a plurality of tiles forming a layer of tiles with a first side and a

second side;

an interlayer binding the layer of tiles to the composite support

layer; and

a signature layer with a first side and a second side, wherein the first side of the signature layer is bonded to the second side of the layer of tiles.

2. The composite armor system, as claimed in claim 1, wherein the interlayer, comprises:

an elastomer layer with a first side and a second side with the first side of the elastomer layer adjacent to a first side of the composite support layer;

a first adhesive layer binding the first side of the composite support layer to the first side of the elastomer layer;

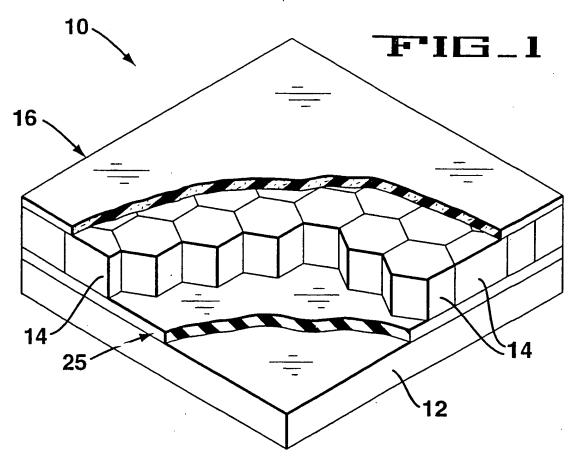
a second adhesive layer binding the second side of the elastomer layer with the first side of the layer of tiles; and

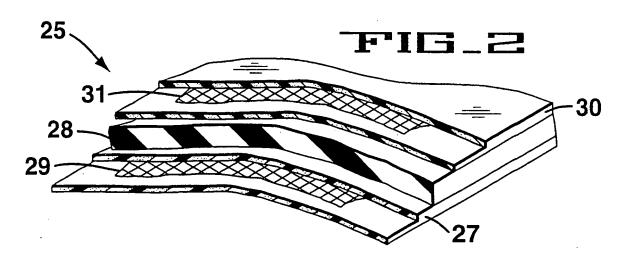
3. The composite armor system, as claimed in claim 2, wherein the signature layer, comprises:

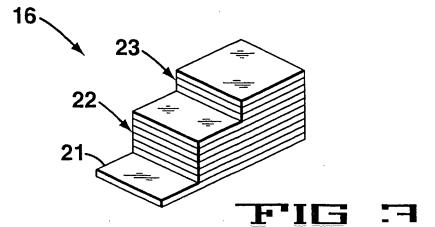
a ground plane scrim with a first side and a second side, wherein the first side of the ground plane is bonded to the second side of the layer of tiles; and

a first plurality of layers of fiberglass impregnated with an epoxy with a first side and a second side, wherein the first side of the first plurality of layers of fiberglass are bonded to the second side of the ground plane scrim.

4. The composite armor system, as claimed in claim 3, wherein the signature layer, further comprises, a second plurality of layers of fiberglass with a first side and a second side, wherein the first side of the second plurality of layers of fiberglass is bonded to the second side of the first plurality of layers of fiberglass.







INTERNATIONAL SEARCH REFURT

Inte onal Application No PCT/US 98/03128

A. CLASSIFICATION OF SUBJECT MATTER IPC 6 F41H5/04 According to International Patent Classification (IPC) or to both national classification and IPC B. FIELDS SEARCHED Minimum documentation searched (classification system followed by classification symbols) Documentation searched other than minimum documentation to the extent that such documents are included in the fields searched Electronic data base consulted during the international search (name of data base and, where practical, search terms used) C. DOCUMENTS CONSIDERED TO BE RELEVANT Citation of document, with indication, where appropriate, of the relevant passages Relevant to claim No. χ WO 91 07633 A (ALLIED SIGNAL) 30 May 1991 1,2 see page 28, line 12 - page 31, line 36 see page 19, line 24 - page 21, line 9 see page 8, line 17 - page 11, line 18 see page 6, line 36 - page 7, line 10 see page 4, line 19-25 see page 3, line 17-33 see page 2, line 30 - page 3, line 3 Α 6,10-12,X GB 1 142 689 A (AEROJET GENERAL CORP.) 12 February 1969 see page 3, line 70 - page 4, line 24; figures 3A,3B Υ 2-6, 12 - 18Further documents are listed in the continuation of box C. Patent family members are listed in annex. Special categories of cited documents: "T" later document published after the international filing date or priority date and not in conflict with the application but "A" document defining the general state of the art which is not cited to understand the principle or theory underlying the considered to be of particular relevance invention "E" earlier document but published on or after the international "X" document of particular relevance; the claimed invention filing date cannot be considered novel or cannot be considered to document which may throw doubts on priority claim(s) or involve an inventive step when the document is taken alone which is cited to establish the publication date of another "Y" document of particular relevance; the claimed invention citation or other special reason (as specified) cannot be considered to involve an inventive step when the "O" document referring to an oral disclosure, use, exhibition or document is combined with one or more other such documents, such combination being obvious to a person skilled document published prior to the international filing date but later than the priority date claimed "&" document member of the same patent family Date of the actual completion of theinternational search Date of mailing of the international search report 13 May 1998 22/05/1998 Name and mailing address of the ISA Authorized officer European Patent Office, P.B. 5818 Patentlaan 2 NL - 2280 HV Rijswijk Tel. (+31-70) 340-2040, Tx. 31 651 epo nl, Van der Plas, J Fax: (+31-70) 340-3016

INTERNATIONAL SEARCH REPORT

Inte onal Application No
PCT/US 98/03128

C.(Continua Category Y	DE 39 09 561 C (DIEHL GMBH & CO) 27 June 1996 see column 2, line 60-65; figure 1	Relevant to claim No. 2-6, 12-18
Y	DE 39 09 561 C (DIEHL GMBH & CO) 27 June 1996	2-6,
	1996	
v I		
•	GB 2 287 836 A (SECR DEFENCE) 27 September 1995 see the whole document	3-6
X	DE 32 28 264 A (APPRICH HARRY) 5 December 1985 see page 17, paragraph 1 see page 14, line 1 - page 15, paragraph 2; figures 2,5,8	18,19
x	US 5 326 606 A (LABOCK JOSEPH) 5 July 1994 see column 4, line 14-22	18
A	DE 23 59 122 C (BLOHM & VOSS) 3 December 1987 see column 4, line -9; figure 2	1
A	US 4 307 140 A (T. DAVIS) 22 December 1981 see page 5, line 45 - page 6, line 43; figure 1 see column 10, line 1-9	8,10,11, 14-17
A	US 4 940 619 A (SMITH JR W NOVIS ET AL) 10 July 1990 see the whole document	1,3,4
A	WO 94 13878 A (SECR DEFENCE BRIT ;ALDERSON JOHN (GB); VICKERS ALAN FREDERICK (GB)) 23 June 1994 see the whole document	1,3,4, 18-20
A	US 5 144 772 A (KAWAMATA TAISUKE ET AL) 8 September 1992 see column 3, line 27-39	1-3
A	FR 1 604 515 A (ELTRO) 29 November 1971 see page 1, line 1-20; figure 1 see page 2, line 8-18 see page 3, line 20 - page 4, line 5	1-3
A	DE 14 28 745 A (L. WESCH) 25 March 1971 see line 8-18 see page 3, line 20 - page 4, line 5	
A	DE 35 08 848 A (MAN TECHNOLOGIE GMBH) 25 September 1986	
Α	DE 14 53 895 A (L. WESCH) 12 August 1971	
A	EP 0 024 713 A (THIELE & CO) 11 March 1981	

Form PCT/ISA/210 (continuation of second sheet) (July 1992)

5

2

INTERNATIONAL SEARCH REPURI

Information on patent family members

inte. onal Application No PCT/US 98/03128

Patent document cited in search repor	nt	Publication date	Patent family member(s)	Publication date
WO 9107633	Α	30-05-1991	CA 2072748 A EP 0493532 A JP 5501604 T	14-05-1991 08-07-1992 25-03-1993
GB 1142689	Α		NONE	
DE 3909561	С	27-06-1996	FR 2733313 A IT 1263161 B	25- 1 0-1996 02-08-1996
GB 2287836	A	27-09-1995	AT 151130 T AU 672074 B AU 5572794 A DE 69309489 D DE 69309489 T EP 0672206 A CA 2150472 A ES 2100035 T WO 9413878 A	15-04-1997 19-09-1996 04-07-1994 07-05-1997 10-07-1997 20-09-1995 23-06-1994 01-06-1997 23-06-1994
DE 3228264	Α	05-12-1985	FR 2573193 A GB 2149482 A,B	16-05-1986 12-06-1985
US 5326606	А	05-07-1994	NONE	
DE 2359122	С	03-12-1987	NONE	
US 4307140	Α	22-12-1981	NONE	
US 4940619	Α	10-07-1990	NONE	
WO 9413878	A	23-06-1994	AT 151130 T AU 672074 B AU 5572794 A CA 2150472 A DE 69309489 D DE 69309489 T EP 0672206 A ES 2100035 T GB 2287836 A,B	15-04-1997 19-09-1996 04-07-1994 23-06-1994 07-05-1997 10-07-1997 20-09-1995 01-06-1997 27-09-1995

INTERNATIONAL SEARCH REPORT

information on patent family members

Inter nal Application No PCT/US 98/03128

Patent document cited in search report		Publication date	Patent family member(s)		Publication date
US 5144772	A	08-09-1992	JP JP EP	2292147 A 2686141 B 0397049 A	03-12-1990 08-12-1997 14-11-1990
FR 1604515	Α	29-11-1971	DE DE	1453894 A 1453895 A	12-08-1971 12-08-1971
DE 1428745	Α	25-03-1971	NONE		
DE 3508848	Α	25-09-1986	NONE		
DE 1453895	Α	12-08-1971	DE FR	1453894 A 1604515 A	12-08-1971 29-11-1971
EP 0024713	A	11-03-1981	DE	2934050 A	26-03-1981

THIS PAGE BLANK (USPTO)